



D8 Series: Experimental Assessment of Boundary Layer Ingestion Benefit

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Backgrounds

- D8 series aircraft configuration offers potentially large efficiency benefit
- Experimental assessment needed to address issues with configuration
 - Unconventional design traits: "double-bubble" fuselage, boundary layer ingestion
 - Potential challenges: engine response to distortion, tail configuration aerodynamics



Goals

- Experimentally assess D8 series configuration performance
- Compare D8 series performance to baseline (737-800) in a traceable manner
 - Benefits of boundary layer ingestion (BLI)
 - Effect of inlet distortion on fan performance
 - Presence of unanticipated losses due to propulsion system-airframe integration

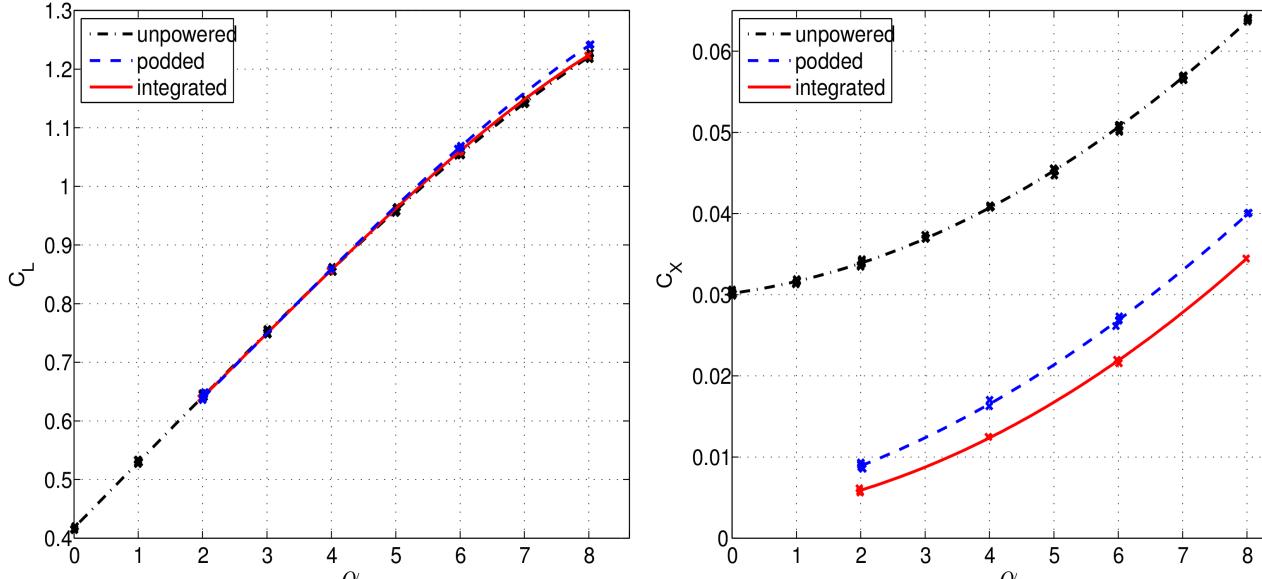


1:20 Unpowered model in MIT Wright Brothers Wind Tunnel (2011) and 1:11 Integrated model NASA Langley Research Center 14'x22' Wind Tunnel (2013)

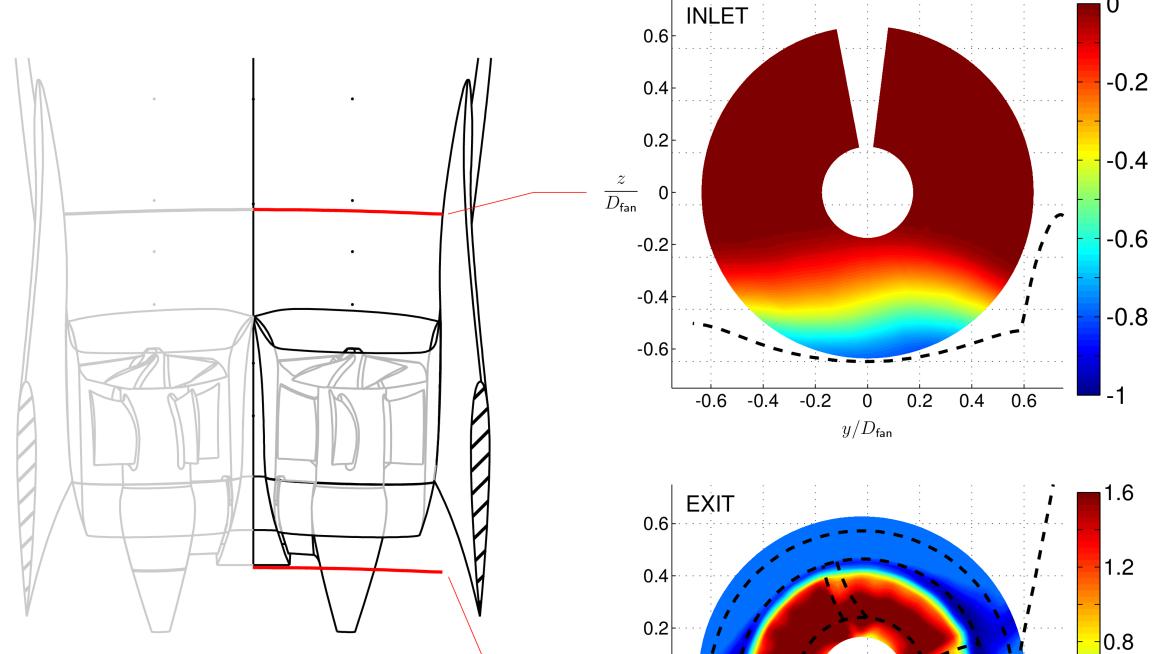
1:11 Powered Airframe Experiments

Aerodynamic curves for the D8-series aircraft

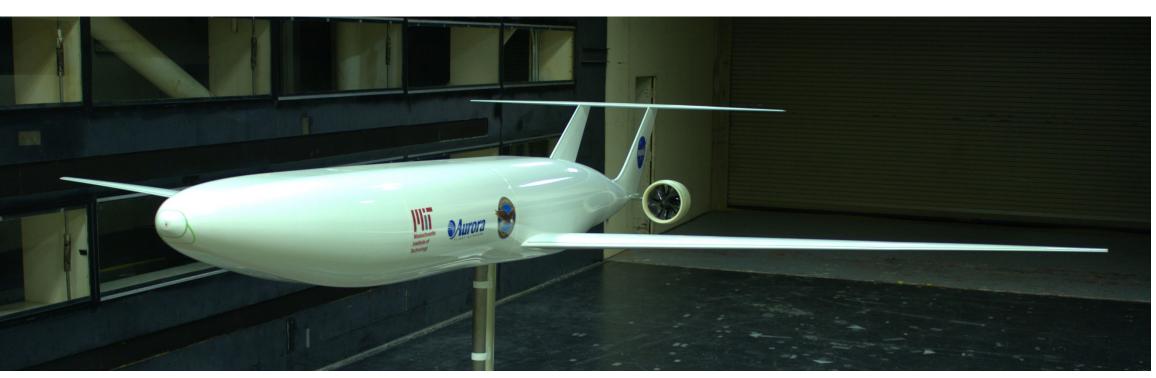
- Purpose: back-to-back comparison of podded propulsor (non-BLI) and integrated (BLI) configurations to assess BLI benefit
- Experimental setup: propulsor power sweeps, flow field surveys
- Experiments were carried out using a commercial off the shelf fans
- Measurements: forces and moments, propulsor mechanical energy flow rates



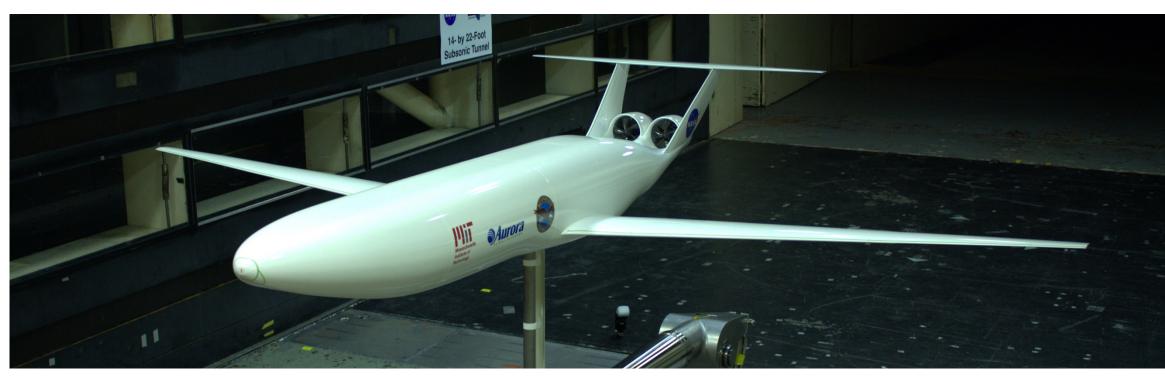
Experimental Total Pressure Surveys (Right Propulsor)



Integrated Configuration $V_{\infty} = 70 \text{ mph}$ $\Omega = 11,000 \text{ RPM}$ $\alpha = 2^{\circ}, \beta = 0^{\circ}$ $\begin{array}{c} \begin{array}{c} z \\ D_{\text{fan}} \\ -0.2 \\ -0.4 \\ -0.6 \\ \hline \\ -0.6 \\ \hline \\ -0.6 \\ -0.6 \\ -0.4 \\ -0.2 \\ 0 \\ 0 \\ -0.2 \\ 0 \\ 0.2 \\ 0.4 \\ 0.6 \\ \hline \\ \\ y/D_{\text{fan}} \end{array} \right) = 0.4$



Podded (non-BLI) configuration



Integrated (BLI) configuration



Boundary Layer Ingestion Benefit

